### Homework 2

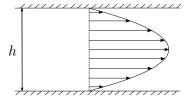
# Analytical solutions of the Navier–Stokes equations

## due February 4, 2016, 12:00

#### Task 1

An aeroplane is flying with a trajectory given by  $r_i(x_i^0, t)$ , where  $t \ge 0$  is the time passed, and  $x_i^0$  the initial position in space. It measures the temperature field  $T(x_i, t)$  in the surrounding air with a thermometer fixed to the wing. Find an expression for the rate of change of the measured temperature. Explain the physical meaning of the terms and compare the expression with that of the material derivative.

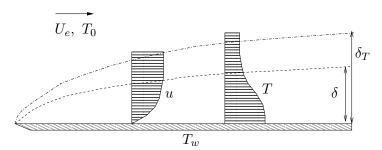
#### Task 2



Consider the fluid flow between two fixed infinite parallel plates, where a pressure gradient  $\frac{\partial p}{\partial x} = C$  is imposed in the direction parallel to the plates. Find a steady solution to the incompressible Navier–Stokes equations for this flow case, *i.e.* so-called plane Poiseuille flow.

(Hint: Assume that the flow is fully developed, meaning  $\frac{\partial}{\partial x} = 0$  for all velocity components and that body forces are absent)

Task 3



Consider the incompressible flow with free-stream velocity  $U_e$  over a semi-infinite flat plate. Let the incoming flow have a temperature  $T_0$  while the plate is kept at the greater temperature  $T_w$ . The temperature field around the plate obeys the following equation:

$$\rho c_p \frac{DT}{Dt} = \kappa \nabla^2 T$$

where  $\kappa$  is the thermal conductivity,  $\rho$  the density and  $c_p$  the specific heat at constant pressure. Assume that the Péclet number, Pe = RePr, is large so that a thin *thermal* boundary layer develops on the plate.

Derive the appropriate boundary-layer approximation of the temperature equation in the limit of large Péclet number and estimate the thermal boundary-layer thickness  $\delta_T$ .

(Hint: Assume a steady flow and that the ratio between the boundary layer thicknesses of flow velocity and temperature scales as  $\delta/\delta_T \sim \mathcal{O}(1)$ )