



AF2903 Road Construction and Maintenance

Royal Institute of Technology, Stockholm

Top-Down Fatigue Cracking Initiation & Propagation

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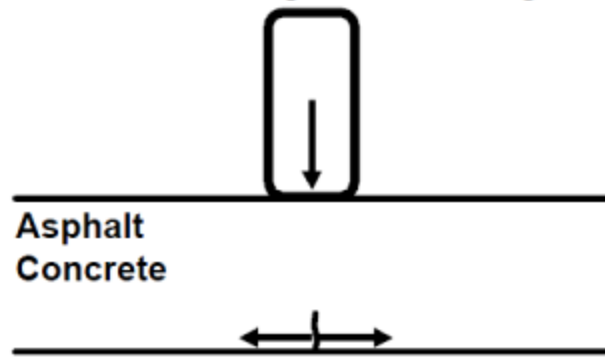
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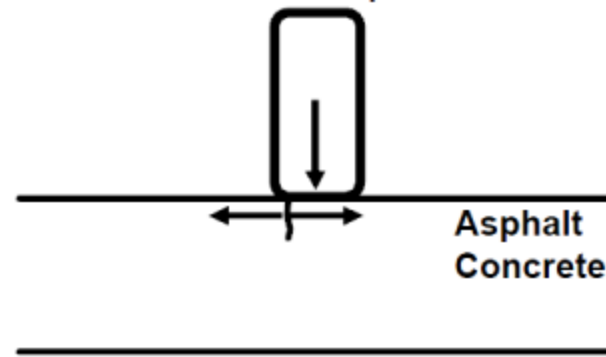
Top-Down Fatigue Cracking

- Top-down fatigue cracking (cracking that initiates at the surface and propagates downwards) has been recognized recently as a major pavement distress
- This cracking phenomena have been observed in many parts of the world and can not be explained by traditional fatigue mechanisms

Traditional Fatigue Cracking



Top-Down Cracking



- We need a model to predict top-down fatigue cracking : Initiation & Propagation

Top-Down Fatigue Cracking



Field core

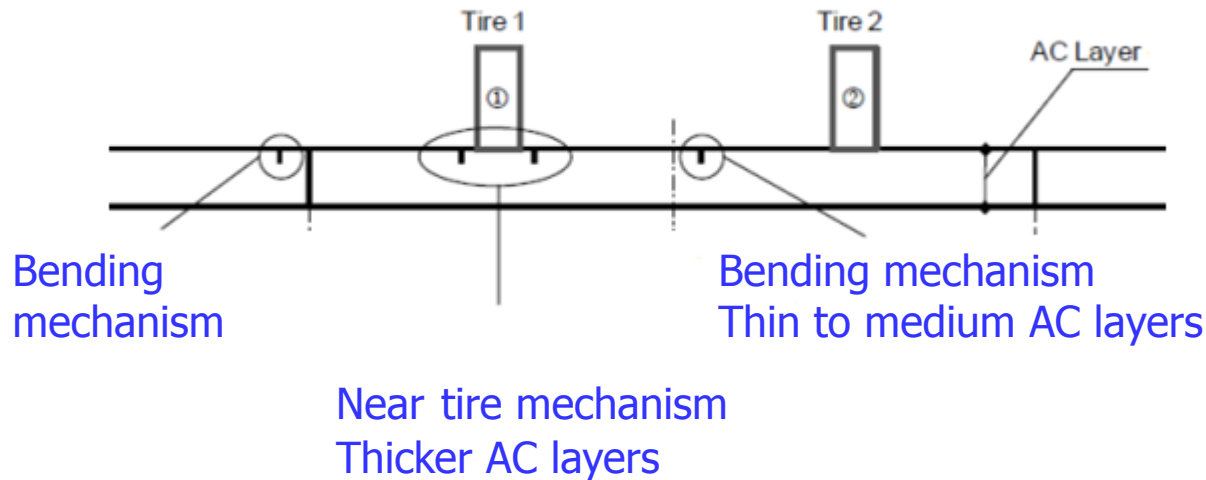


Field section

Top Down Fatigue Cracking Prediction

- **Crack initiation**

- Bending mechanism (bending induced surface tension away from the tire)
- Near tire mechanism (shear induced tension at the tire edge)



- **Crack propagation**

- The presence of cracks: redistribution and intensification of stresses
- Presence of stiffness gradients
 - Temperature variations with in the AC layer
 - Non-uniform aging with in the AC mixture

Unified Predictive System

- **Crack initiation** (Viscoelastic continuum damage (VECD) model)
 - Predict damage zone effects and Identify the time and location of crack initiation
 - Incorporates material property sub-models (aging, healing, failure criteria, viscoelasticity, thermal stress....)
 - Finite element based
- **Crack propagation** (HMA-fracture mechanics (HMA-FM) model)
 - Account effect of macro cracks and predict propagation of crack over time
 - Incorporates material property sub-models (aging, healing, failure criteria, thermal stress..)

Note! Much significant development is needed before implementation into MEPDG to predict top down cracking



Top-Down Fatigue Cracking

Design based on Energy Ratio

(First Generation)

University of Florida (2006), Report No. 0003932

Top-Down Fatigue Cracking Design based on Energy Ratio

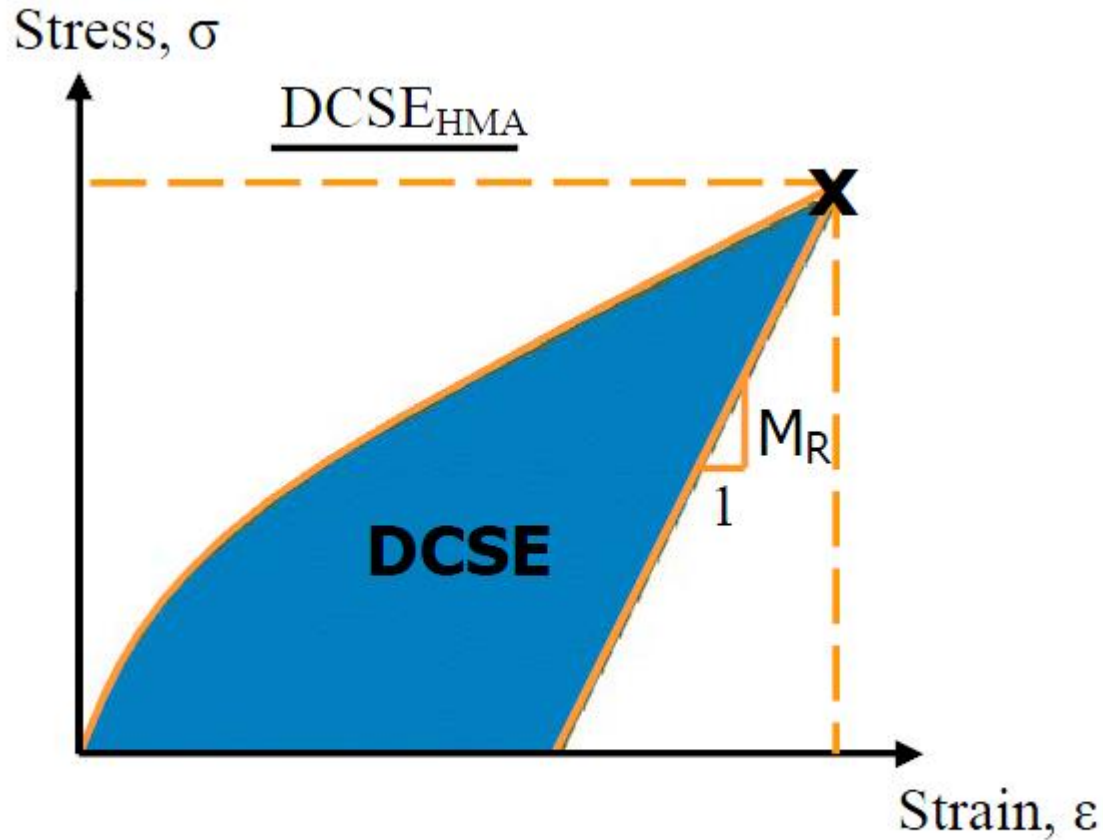
- Fracture mechanics based M-E design method
- Optimize the AC thickness against top-down fatigue cracking

Key features

- Damage equal to the accumulated dissipated creep strain energy ($DCSE_{min}$)
- Damage threshold ($DCSE_{lim}$) independent of loading or loading history
- Damage under the cracking threshold is fully healable
- Macro Crack will initiate if accumulated $DCSE_{min}$ exceeds $DCSE_{lim}$
- Consider only load induced stresses
- Structure and mixture for “averaged” environmental conditions

Dissipated Creep Strain Energy (DCSE)

- Defines a fracture damage threshold



$$DCSE_{HMA} = \text{AREA}$$

Energy Ratio (ER)

- A dimensionless parameter which compares mixture DCSE threshold to that of accumulated load induced DCSE

$$ER = \frac{DCSE_{lim}}{DCSE_{min}}$$

- Mixture properties are needed to determine $DCSE_{lim}$

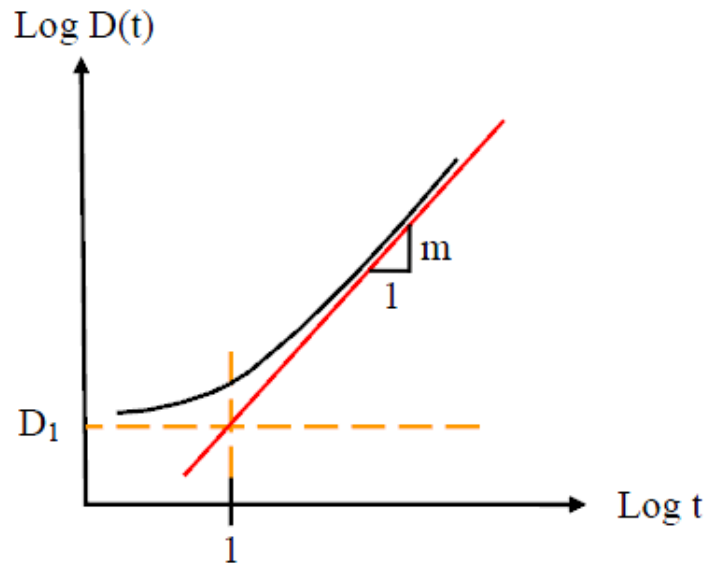
$$DCSE_{lim} = c_f * S_t \frac{mD_1}{10^{3(1-m)}}$$

$$c_f = 6.9 * 10^7$$

S_t tensile strength at the surface of the AC layer

$$D(t) = D_0 + D_1 * t^m,$$

D_1 and m are creep parameters

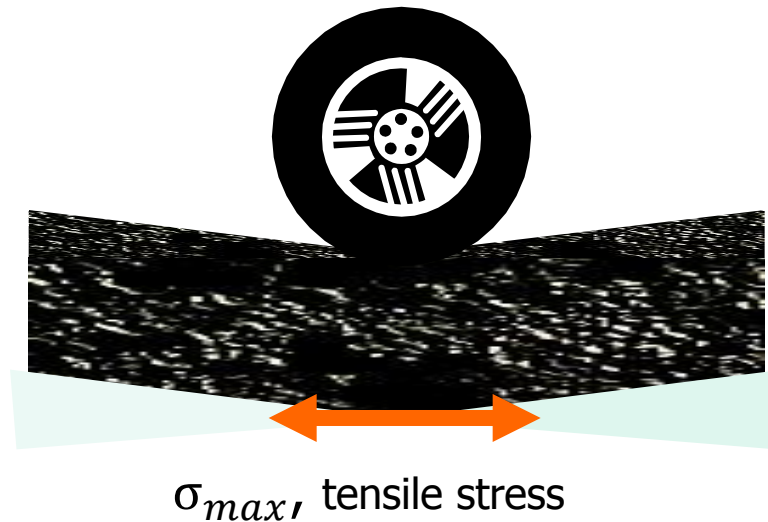


Energy Ratio (ER)

- Structural and mixture properties are needed to determine $DCSE_{min}$

$$DCSE_{min} = m^{2.98} * D_1 / f(S_t, \sigma_{max})$$

$$f(S_t, \sigma_{max}) = 0.0299 * \sigma_{max}^{-3.10} * (6.36 - S_t) + 2.46 * 10^{-8}$$





Design and Analysis

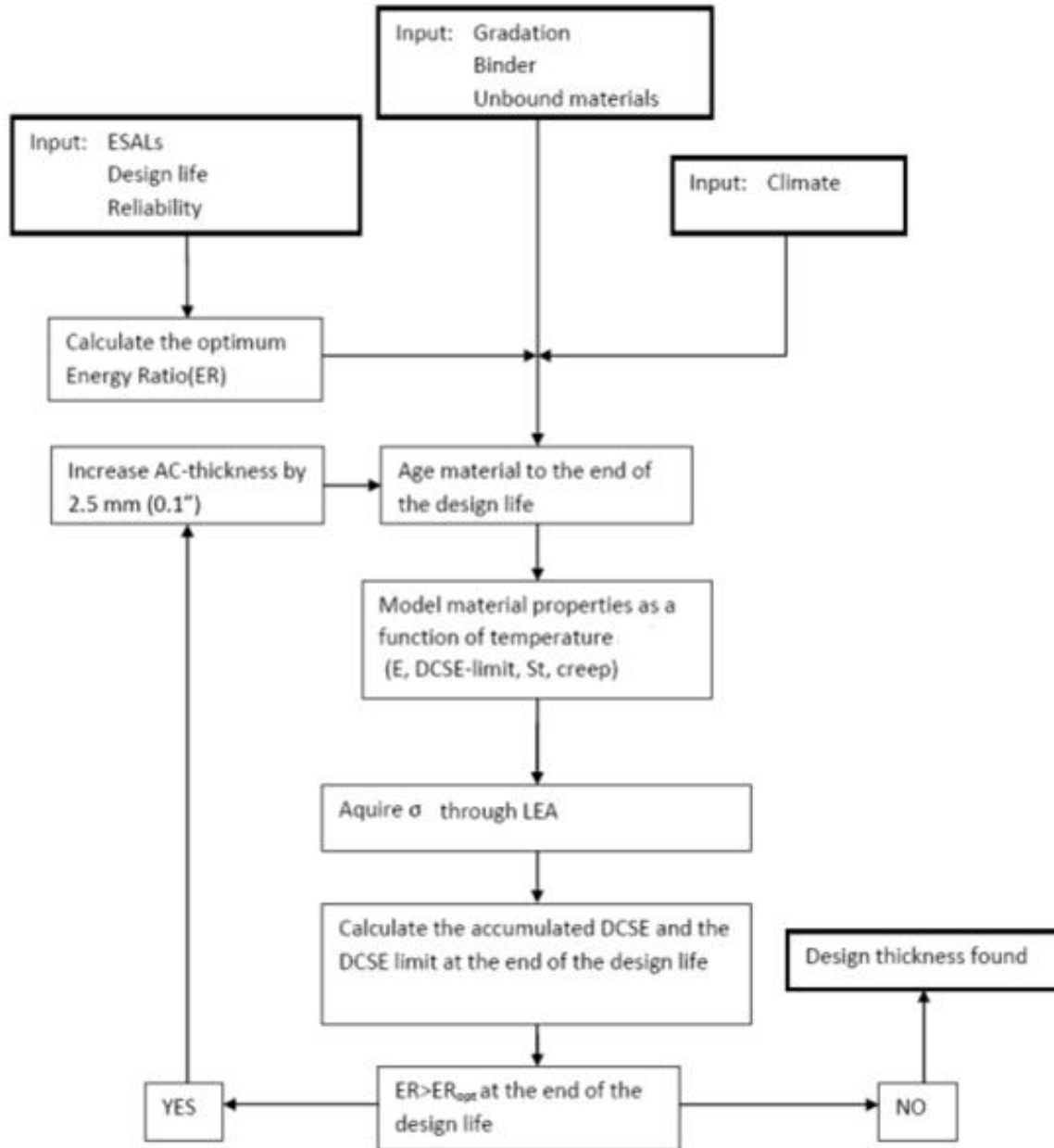
- **Input data**

- Gradation ($\rho_{3/4}$, $\rho_{3/8}$, ρ_4 , ρ_{200})
- Volumetric (V_a , V_{beff})
- Rheological property (A , VTS)
- Traffic volume (**ESALS**) & loading
- Reliability (R)
- Climate (*maat*)
- Design life (*months*)
- Reference temperature

- **Output data**

- Material properties at a given depth (Viscosity, AC stiffness, tensile strength & creep parameters)
- Optimized AC thickness

Flow Chart



Binder Aging Model

- Based on Global Aging Model (GAM) at a reference temperature of 10C (50F)
- Predicts binder viscosity at a given depth and time

Binder viscosity at mix or laydown condition

$$\log \log (\eta) = A + VTS \times \log (T_R)$$

A & VTS : Regression constants
 T_R : Temperature in Rankine scale
 $T_R = 459.67 + T_F$ (Fahrenheit)

Aged binder viscosity at surface

$$\log \log (\eta_{aged}) = F_{AV} \times \frac{\log \log (\eta_{t=0}) + A_f t}{1 + B_f t}$$

$F_{AV} = 1$
 A_f & B_f : field aging parameters
 t : time in months

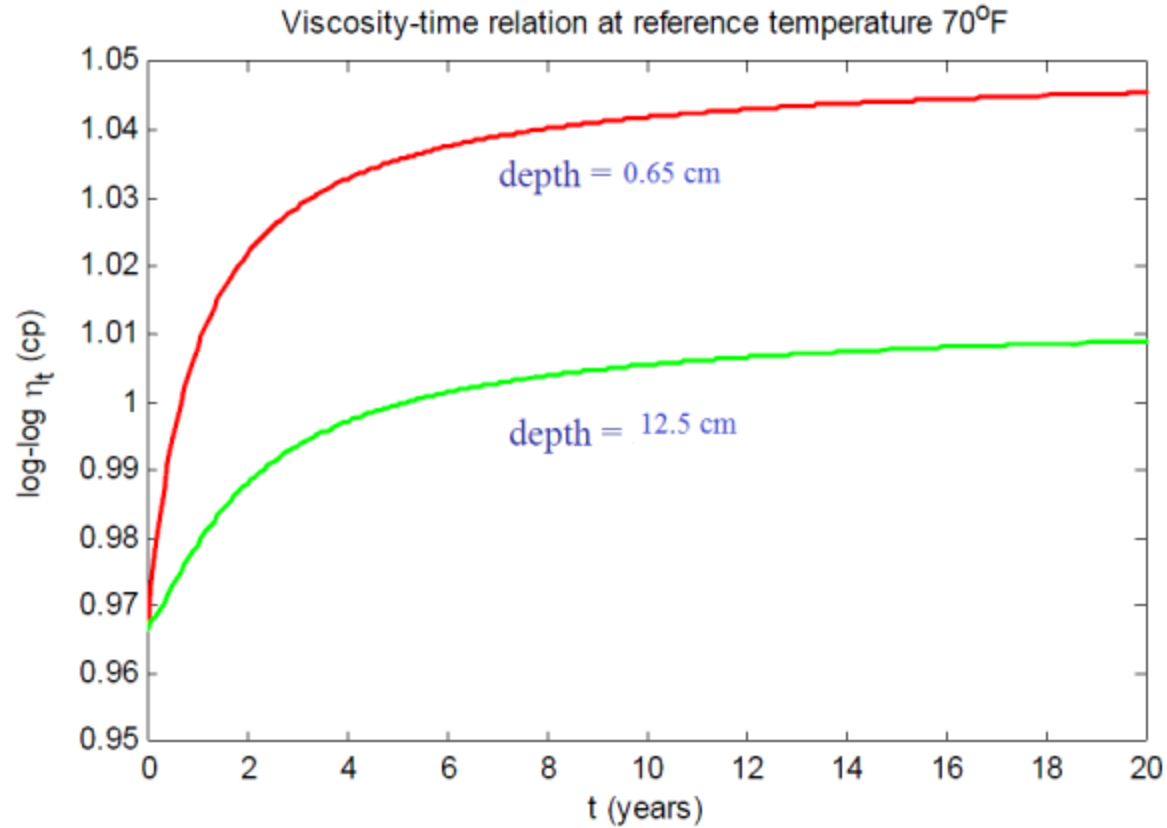
Aged binder viscosity at a given depth and time

$$\eta_{t,z} = \frac{\eta_t (4 + E) - E(\eta_{t=0})(1 - 4z)}{4(1 + E \cdot z)}$$

$$E = 23.82 \exp(-0.0308 \cdot Maat)$$

Binder Aging Model

- Predicted viscosity at a reference temperature of 21C (70F)



AC Stiffness Aging Model

- Based on binder aging model and predicts AC stiffness at a give time & depth

$$\log \left| E^* \right|_o = \delta + \frac{\alpha}{1 + \exp(\beta + \gamma \log t_r)}$$

$\delta, \alpha, \beta, \gamma$ are fitting parameters.

$t_r = 0.1\text{s}$, time of loading

$$\delta = 2.718879 + 0.079524 \times p_{200} - 0.007294 \times (p_{200})^2 + 0.002085 \times \rho_4 \\ - 0.01293 \times V_a + 0.08541 \frac{V_{be}}{V_{be} + V_a}$$

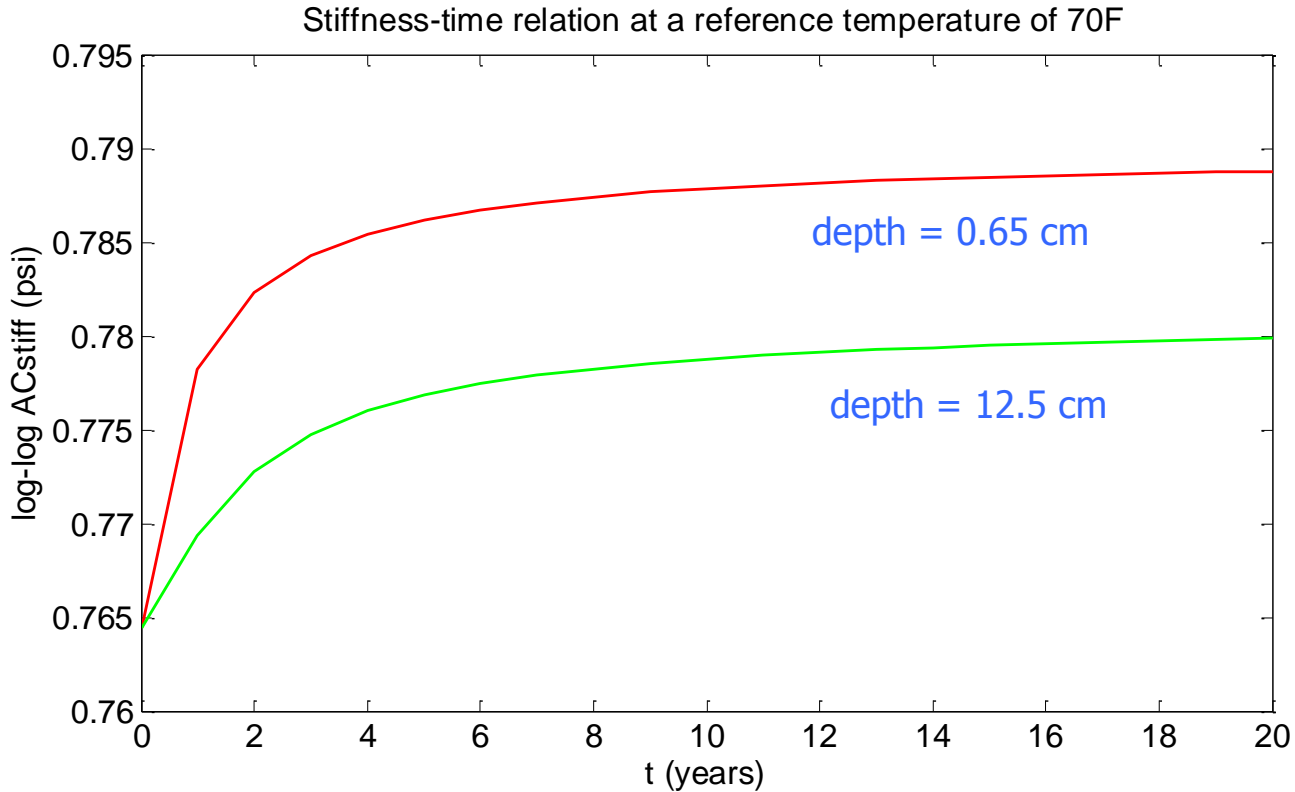
$$\alpha = 3.559267 - 0.005451 \times \rho_4 + 0.020711 \times \rho_{3/8} - 0.000351 \times (\rho_{3/8})^2 \\ + 0.00532 \times \rho_{3/4}$$

$$\beta = -0.513574 - 0.355353 \times \log(\eta_r)$$

$$\gamma = 0.37217$$

AC Stiffness Aging Model

- Predicted AC stiffness at a reference temperature of 21C (70F)



Tensile Strength Aging Model

- Predicts AC tensile strength at a given depth and time
- Based on the stiffness aging model at a loading time of 1800s

Tensile strength (MPa)

$$S_t = \sum_{n=0}^5 a_n \cdot (\log S_f)^n$$

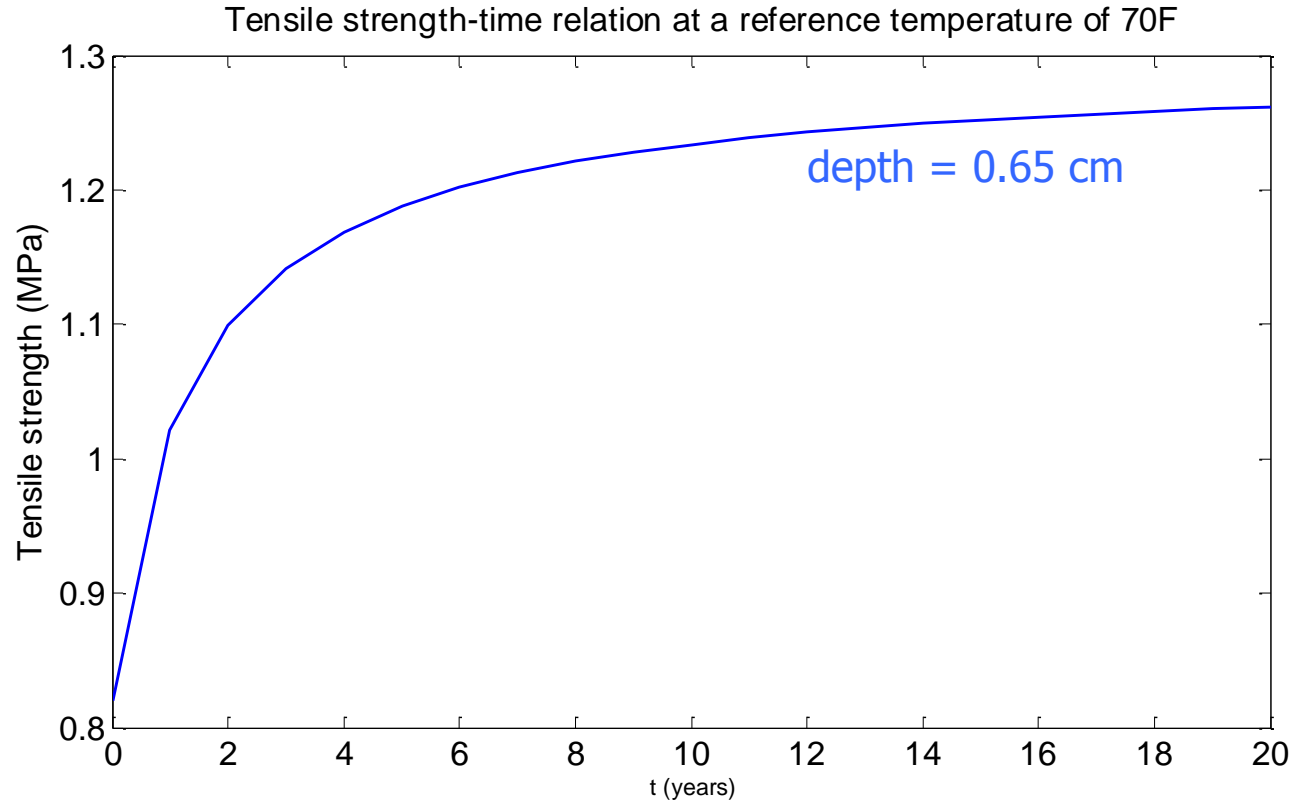
AC stiffness (psi)

- Regression constant (a_n)

$$a_0 = 284.01, \quad a_1 = -330.02, \quad a_2 = 151.02, \quad a_3 = -34.03,$$
$$a_4 = 3.7786, \quad a_5 = -0.1652$$

Tensile Strength Aging Model

- Tensile strength should be calculated near AC surface at 0.65 cm (0.25 inch)



Creep Parameters (m, D_1) aging model

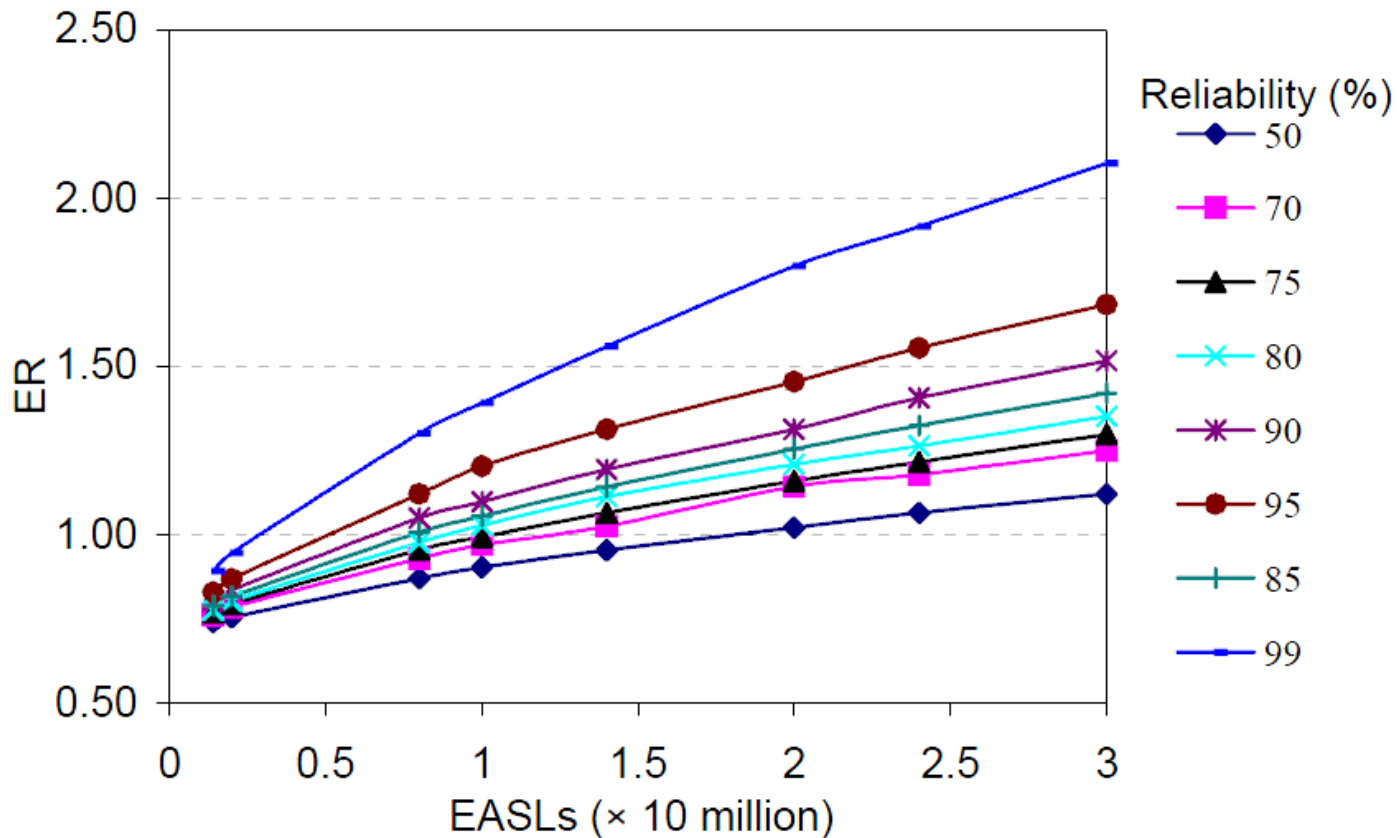
- Gradation, volumetric and binder property of the mix are required
- Creep parameters should be calculated near AC surface 0.65 cm (0.25 inch)

$$\log(D_0) = -\delta - \alpha - \log \lambda_r, \quad \log(D_0 + D_1) = -\delta - \frac{\alpha}{1 + e^\beta} - \log \lambda_r \quad \lambda_r = 0.4$$

$$m_0 = \alpha \gamma \times \frac{\exp(\beta + 3\gamma)}{[1 + \exp(\beta + 3\gamma)]^2} \quad m = m_0 + \frac{\kappa}{\log \log \eta} \quad k = 0.408$$

Energy Ratio Optimum (ER_{opt})

- ER=1 is a reference point, energy ratio lower than 1 leads to weaker pavement
- Higher traffic and higher reliability requires a higher energy ratio



Energy Ratio optimum = $f(\text{ESALS}, \text{Reliability})$

- Based on filed calibration

ER_{opt}

Reliability (%)	$ER_{\text{opt}} = f(x: \text{ESALS in 10 millions})$		
	$x < 0.14$	$0.14 \leq x < 0.8$	$x \geq 0.8$
50	$ER_{\text{opt}} = 0.112 x + 0.65$	$ER_{\text{opt}} = 0.311 x + 0.62$	$ER_{\text{opt}} = 0.114 x + 0.78$
70	$ER_{\text{opt}} = 0.162 x + 0.65$	$ER_{\text{opt}} = 0.388 x + 0.62$	$ER_{\text{opt}} = 0.146 x + 0.81$
75	$ER_{\text{opt}} = 0.192 x + 0.65$	$ER_{\text{opt}} = 0.423 x + 0.62$	$ER_{\text{opt}} = 0.156 x + 0.83$
80	$ER_{\text{opt}} = 0.214 x + 0.65$	$ER_{\text{opt}} = 0.4545 x + 0.62$	$ER_{\text{opt}} = 0.170 x + 0.84$
85	$ER_{\text{opt}} = 0.241 x + 0.65$	$ER_{\text{opt}} = 0.491 x + 0.62$	$ER_{\text{opt}} = 0.187 x + 0.86$
90	$ER_{\text{opt}} = 0.276 x + 0.65$	$ER_{\text{opt}} = 0.546 x + 0.61$	$ER_{\text{opt}} = 0.212 x + 0.88$
95	$ER_{\text{opt}} = 0.329 x + 0.65$	$ER_{\text{opt}} = 0.640 x + 0.61$	$ER_{\text{opt}} = 0.256 x + 0.92$
99	$ER_{\text{opt}} = 0.454 x + 0.65$	$ER_{\text{opt}} = 0.876 x + 0.60$	$ER_{\text{opt}} = 0.365 x + 1.01$

Pavement Thickness Design

- Performed based on the amount of damage at the end of the pavement life and the damage criterion
- Dissipated creep strain energy minimum

$$DCSE_{\min} = m^{2.98} * D_1 / f(S_t, \sigma_{\max})$$

- Dissipated creep strain energy limit

$$DCSE_f = c_f * S_t \frac{mD_1}{10^{3(1-m)}}$$

- Energy ratio

$$ER = \frac{DCSE_{\lim}}{DCSE_{\min}}$$

- Optimum thickness

$$ER_{opt} \approx ER$$



Simplified Cracking Performance Prediction Model

(Second Generation)

NCHRP Project 1-42A (2010)

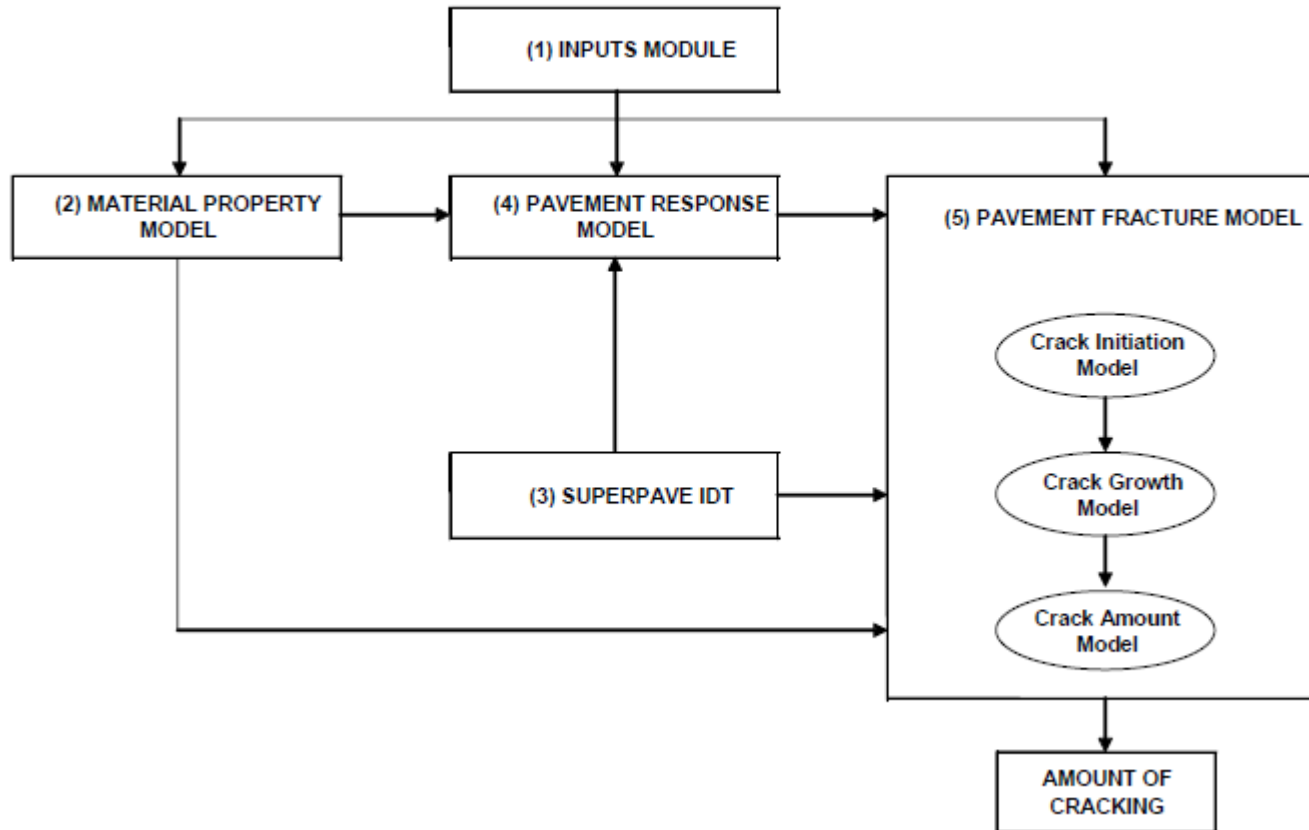


Simplified Cracking Performance Prediction Model

- Uses a simplified fracture energy based approach to predict crack initiation instead of the viscoelastic continuum damage model (VECD)
- Damage zones are not considered
- Fracture mechanics model to predict crack propagation
- **Key elements**
 - A critical condition concept that accurately capture field observations
 - Material property sub-models that account aging of near surface mixture properties
 - Increase in stiffness (stiffening)
 - Reduction in fracture energy (embrittlement)
 - Reduction in healing potential
 - Thermal response model to predict transverse thermal stresses
 - Pavement fracture model that predict crack growth with time

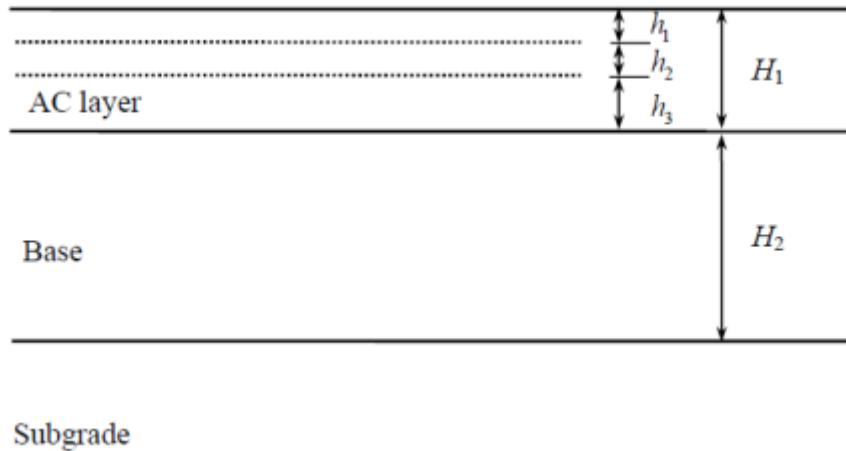
Framework

- The overall framework of the integrated simplified system



Material Property Sub-Model

- Predict material properties



Sub-model	Sub-model component	Input requirement
Material property model	AC stiffness aging model	- Basic mixture characteristics (gradation, binder type, mix volumetrics) - Temperature, loading time, and aging time
	AC tensile strength aging model	- Stiffness (from AC stiffness aging model) - Material coefficients a_n
	Fracture energy limit aging model	- Stiffness (from AC stiffness aging model) - Initial fracture energy
	Healing model	- Aging parameter k_1 (to be determined in calibration) - Stiffness (from AC stiffness aging model) - Initial fracture energy - Critical stiffnesses.

AC-Stiffness Aging Model

- Based on binder aging model & dynamic modulus model (at loading time of $t_r = 0.1\text{s}$)
- Considers stiffness gradient due to temperature & aging by dividing the AC layer into sub-layers

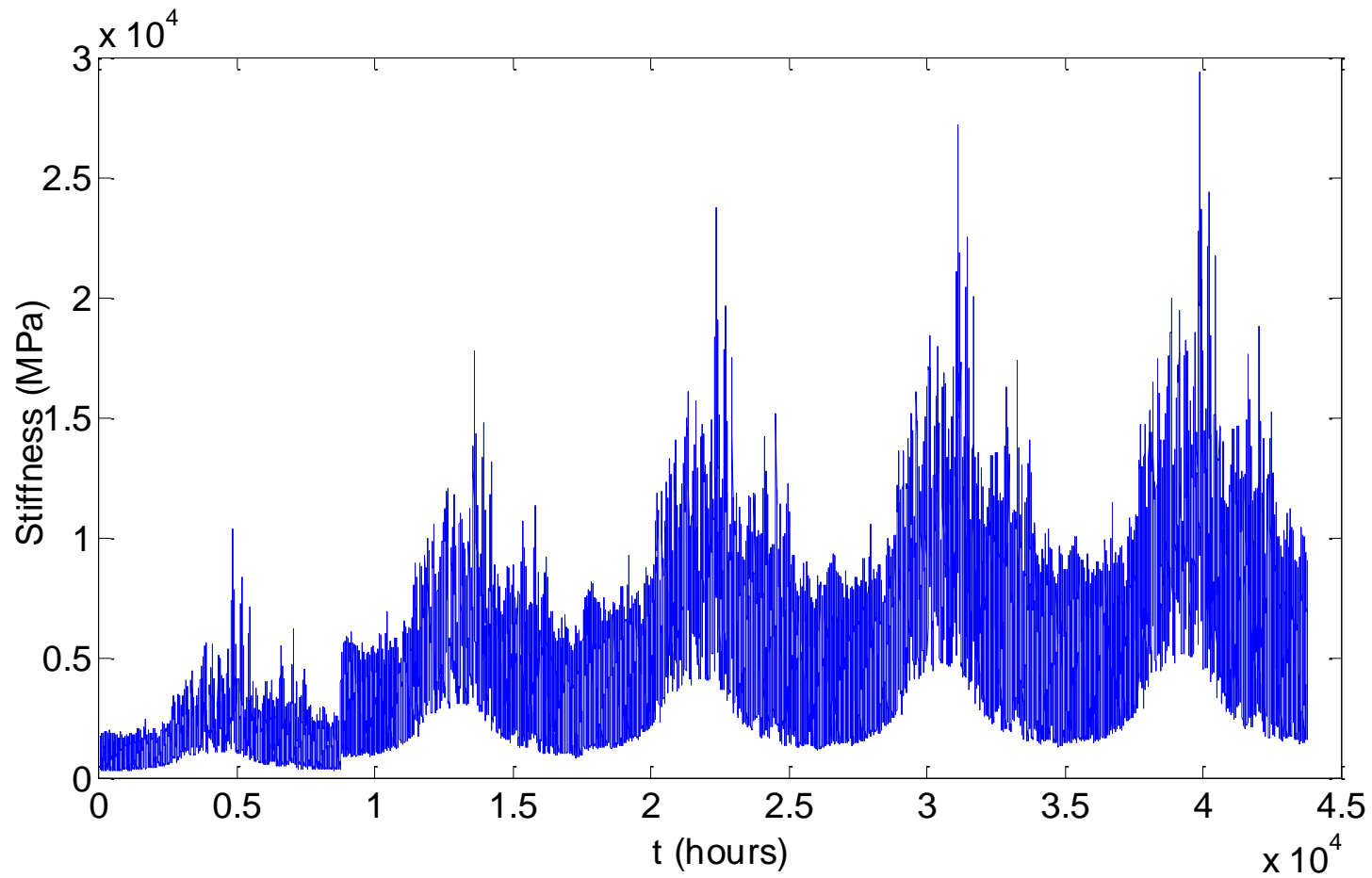
$$\left| E^* \right|_t = \left| E^* \right|_0 \frac{\log \eta_t}{\log \eta_0}$$

$$\eta_{t,z} = \frac{\eta_t(4 + E) - E(\eta_{t=0})(1 - 4z)}{4(1 + E \cdot z)}$$

$$\log \left| E^* \right|_0 = \delta + \frac{\alpha}{1 + \exp(\beta + \gamma \log t_r)}$$

$\delta, \alpha, \beta, \gamma$ are fitting parameters.

Predicted Stiffness (five years)



Inputs for AC Stiffness Aging Model

- **Mix gradation**

$\rho_{3/4}$ – Percent weight retained on 19mm (3/4 inch)sieve

$\rho_{3/8}$ – Percent weight retained on 9.5 mm (3/8 inch) sieve

ρ_4 – Percent weight retained on 4.75 mm (3/8 inch) sieve

ρ_{200} – Percent weight passing 0.75 mm sieve

- **Mix volumetric**

V_a – Percent air void content by volume

V_{be} – Effective asphalt content, by percent

- **Mix rheological property (Binder type – PG)**

A & VTS – Regression constants

PG =67-22: A=10.6316 VTS =-3.548

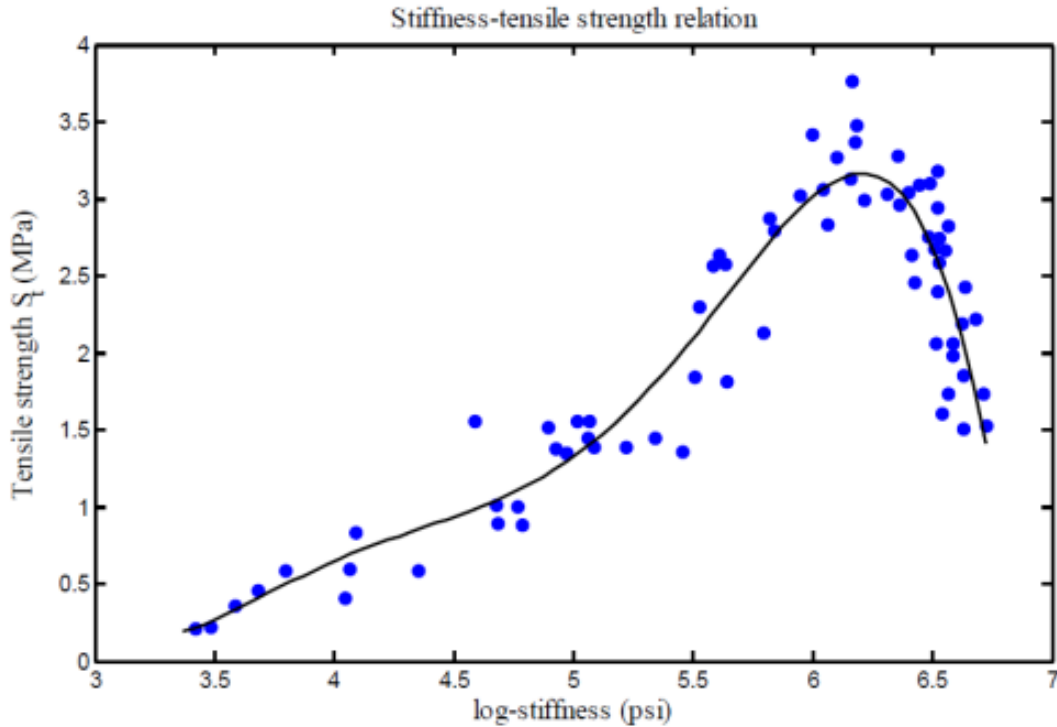
PG=76-22: A=9.715 VTS =-3.208

Tensile Strength Aging Model

- Based on AC- stiffness aging model (at a loading time of 1800s)

$$S_t = \sum_{n=0}^5 a_n \cdot (\log S_f)^n$$

Unit: S_f (psi), S_t (MPa)



Fracture Energy Limit Aging Model

- Fracture energy limit decreases with age and reach some minimum value of (FE_{\min}) = 0.2kJ/m³ at the 50th year

$$FE_f(t) = FE_i - (FE_i - FE_{\min}) \cdot [S_n(t)]^{k_1}$$

$$S_n(t) = \frac{S(t) - S_0}{S_{\max} - S_0}$$

Normalized
stiffness at AC layer

FE_i and k_1 can be determined from IDT test in the lab

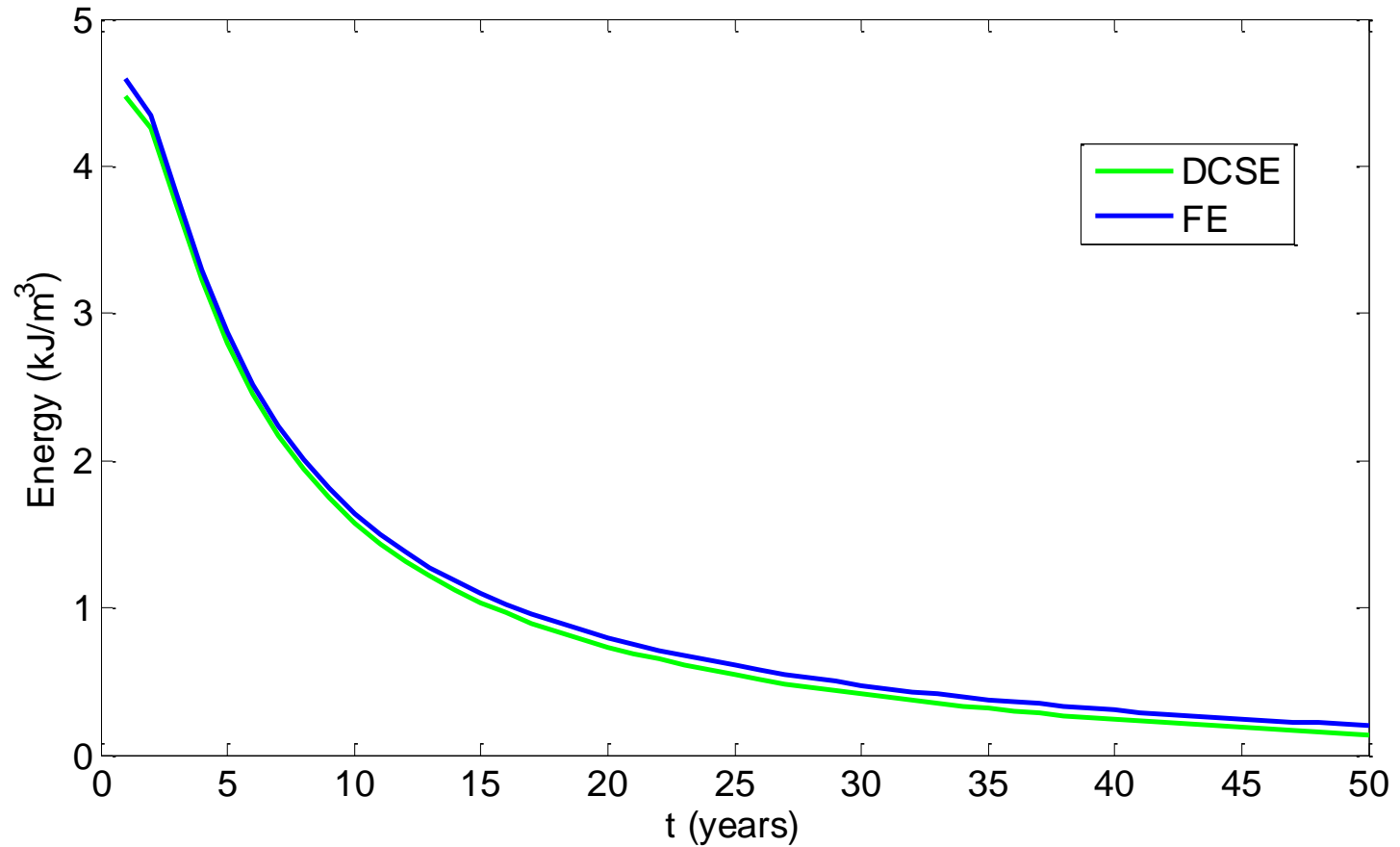
- DCSE limit aging function is developed based on FE limit aging model

AC stiffness

$$DCSE_f(t, z) = FE_f(t, z) - [S_t(t, z)]^2 / [2 \cdot S(t, z)]$$

Tensile
strength

Fracture Energy Limit Aging Model



$$FE = DCSE + EE \text{ (Elastic Energy)}$$

Healing Model

- All damage does not cause crack some will heal
- Composed of three components
 - Maximum healing potential aging model

$$h_{ym}(t) = 1 - [S_n(t)]^{FE_i/1.67}$$

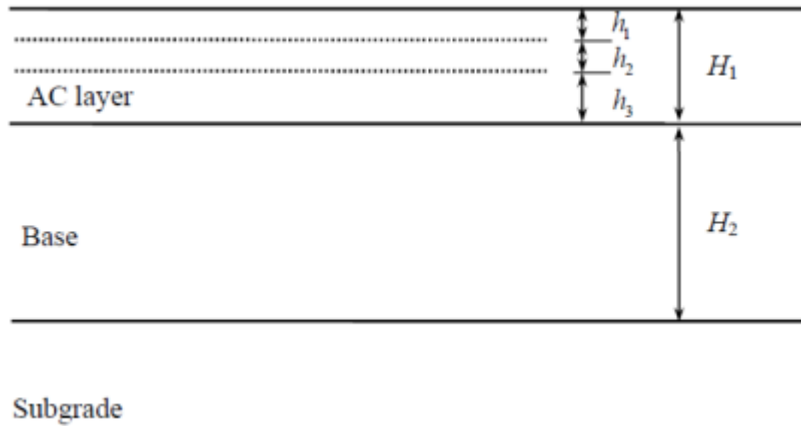
- Daily based healing criterion to estimate the daily recovered damage

$$DCSE_{d_remain} = DCSE_{d_induced} \cdot (1 - h_{dn})$$

- Yearly based healing criterion to estimate continuous healing

$$DCSE_{y_remain} = DCSE_{y_induced} \cdot (1 - h_{ym})$$

Pavement Response Sub-Model

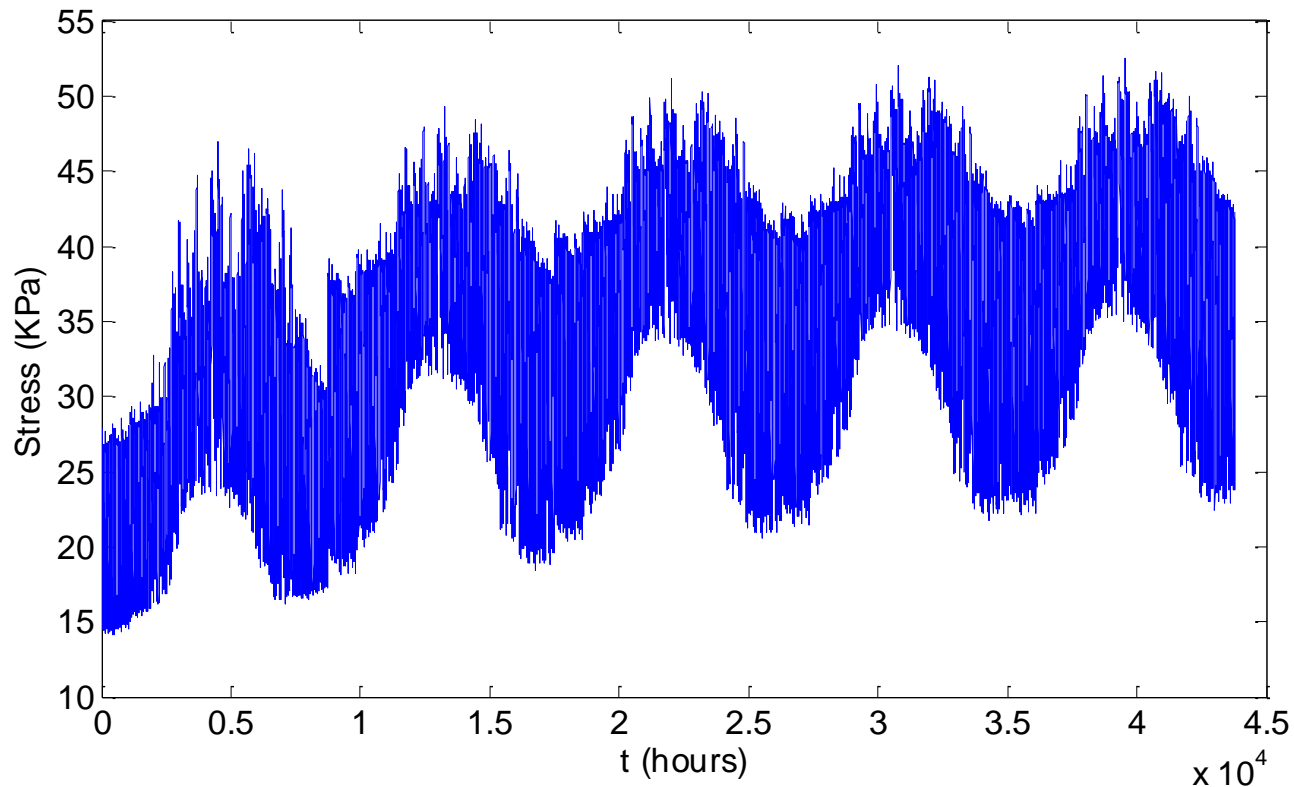
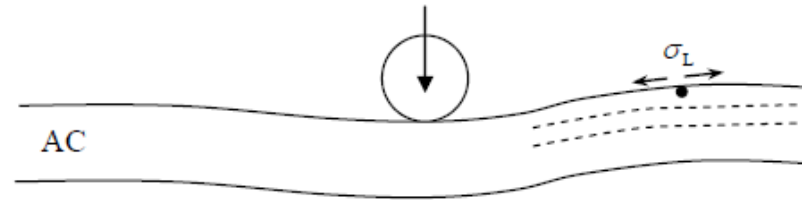


Sub-model	Sub-model component	Input requirement
Pavement response model	Load response model	<ul style="list-style-type: none"> - Structural properties of each layer (thickness, modulus, and Poisson's ratio) - Stiffness (from AC stiffness aging model) - Equivalent single axle load
	Thermal response model	<ul style="list-style-type: none"> - Structural property of AC layer (thickness) - Relaxation modulus master curve parameters: E_i, λ_i, η_v - Temperature and thermal contraction coefficient

Load Response Model

- Predict maximum surface tensile stress due to a circular load using a 3D linear elastic analysis (LEA)

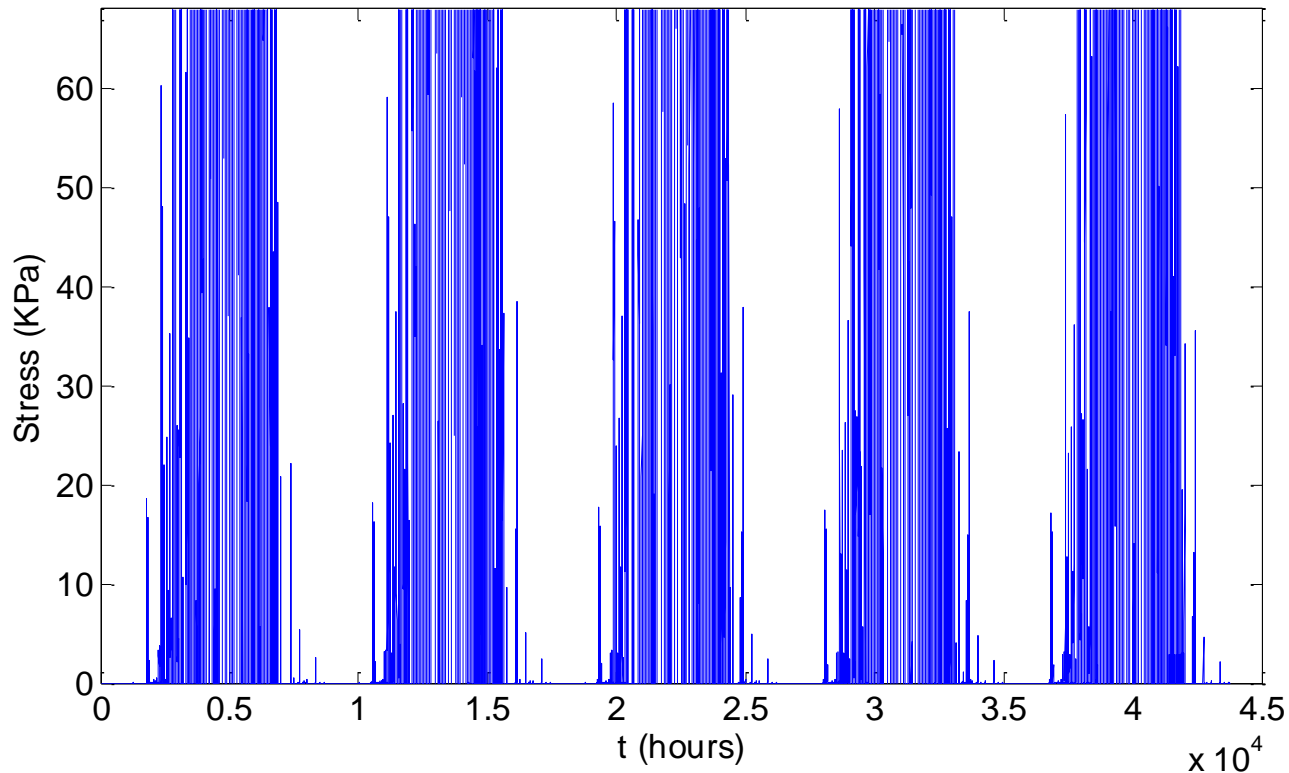
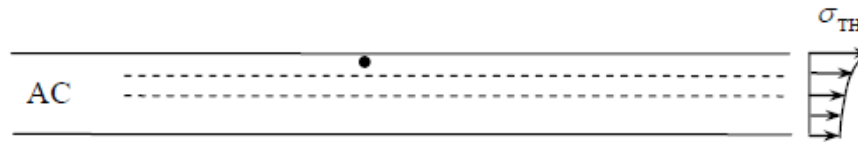
Load response:



Thermal Response Model

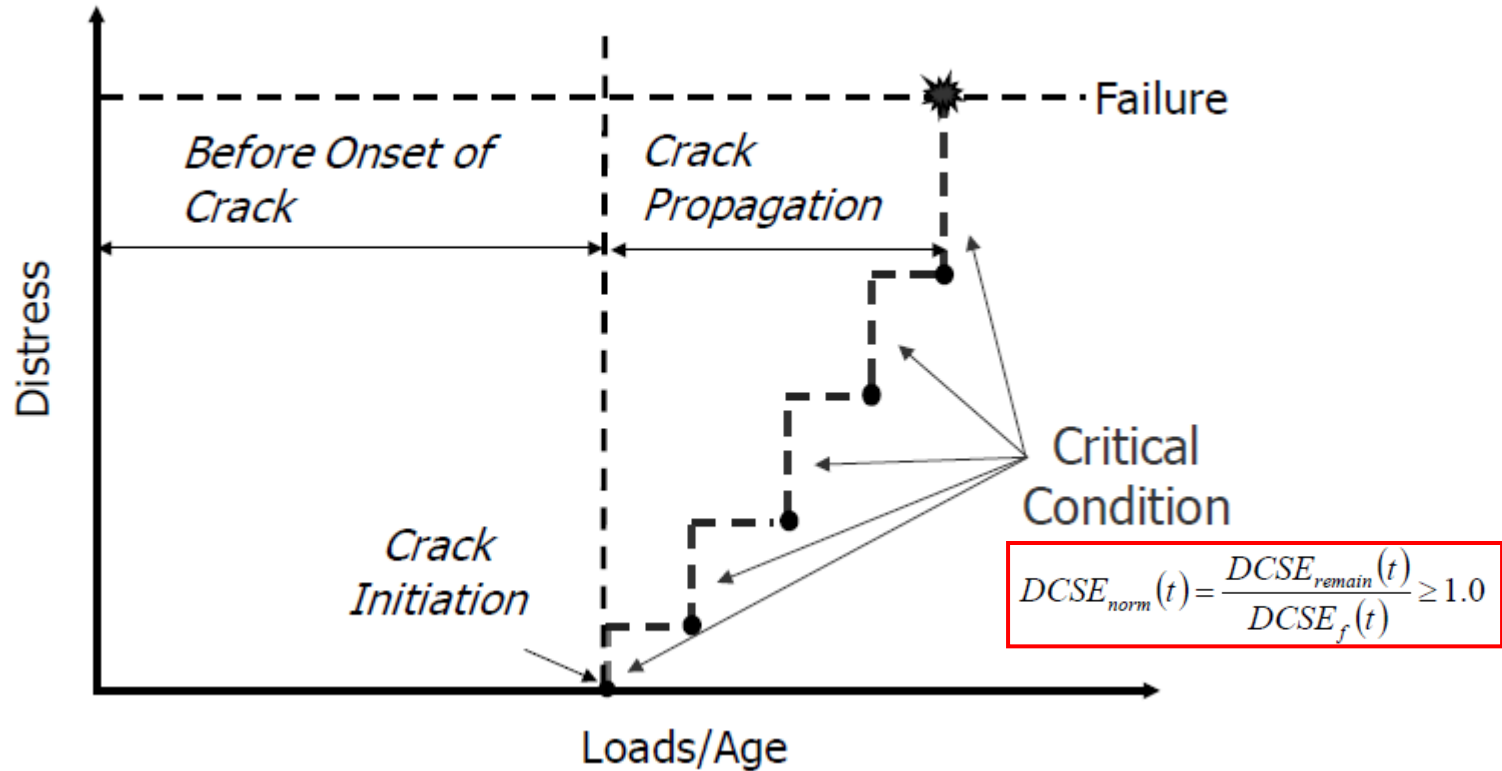
- Predict thermally induced transverse stress
- Maximum of 10psi (68.9KPa) as it can not exceed the friction limit for typical HMA & base materials

Thermal response:



Crack Initiation Model

- Developed on the basis of a threshold concept
- Predicts location and time of crack initiation



Load Associate Damage

Load induced stress

$$DCSE_L / cycle = \int_0^{0.1} \sigma_{AVE} \sin(10\pi t) \dot{\epsilon}_{pmax} \sin(10\pi t) dt$$

Creep strain rate

$$\epsilon_{pmax} = m * D_1 * (1000)^{(m-1)}$$

- Gradation, volumetric and binder property of the mix are required to calculate creep parameters (m, D1)

$$\log(D_0) = -\delta - \alpha - \log \lambda_r, \quad \log(D_0 + D_1) = -\delta - \frac{\alpha}{1 + e^\beta} - \log \lambda_r$$

$$m_0 = \alpha \gamma \times \frac{\exp(\beta + 3\gamma)}{[1 + \exp(\beta + 3\gamma)]^2}$$

$$m = m_0 + \frac{\kappa}{\log \log \eta}$$

Thermal Associated Damage

$$DCSE_T / \Delta t = [\sigma(t) - \sigma(t - \Delta t)] \cdot [\varepsilon_{cr}(t) - \varepsilon_{cr}(t - \Delta t)] / 2$$

$$\varepsilon_{cr} = m * D_1 * (1000)^{(m-1)} * 3600$$

- Gradation, volumetric and binder property of the mix are required to calculate creep parameters (m, D1)

$$\log(D_0) = -\delta - \alpha - \log \lambda_r, \quad \log(D_0 + D_1) = -\delta - \frac{\alpha}{1 + e^\beta} - \log \lambda_r$$

$$m_0 = \alpha \gamma \times \frac{\exp(\beta + 3\gamma)}{[1 + \exp(\beta + 3\gamma)]^2}$$

$$m = m_0 + \frac{\kappa}{\log \log \eta}$$

Crack Initiation Model

- Dissipated Creep strain Energy Limit ($DCSE_{lim}$)

$$DCSE_f(t, z) = FE_f(t, z) - [S_t(t, z)]^2 / [2 \cdot S(t, z)]$$

- Dissipated Creep strain Energy remaining after healing ($DCSE_{remain}$)

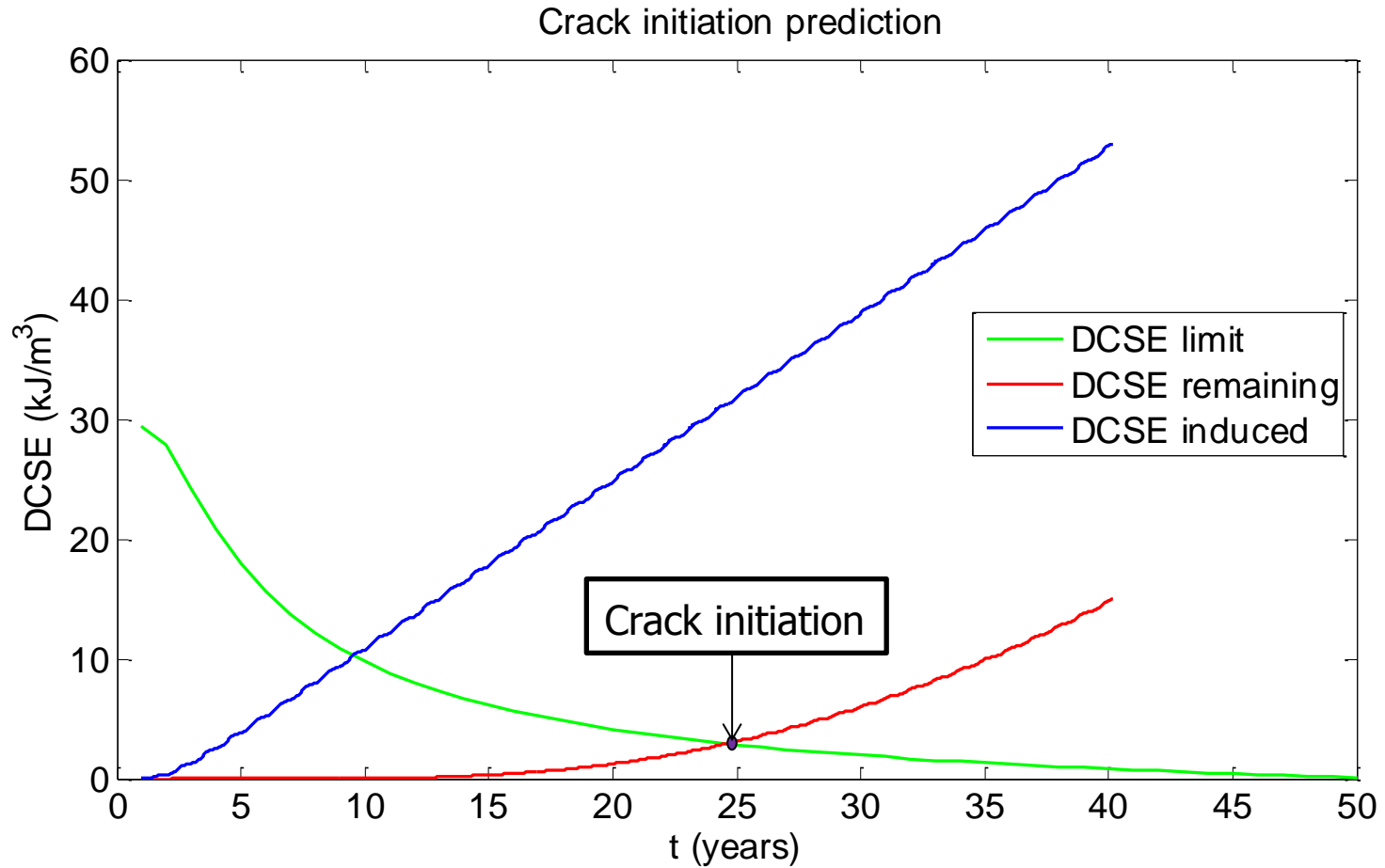
$$DCSE_{remain}(\Delta t) = (1 - h_{dn}) \cdot [n \cdot (DCSE_L / cycle) + DCSE_T(\Delta t)]$$

n is number of load cycles in Δt

- Crack initiation

$$DCSE_{norm}(t) = \frac{DCSE_{remain}(t)}{DCSE_f(t)} \geq 1.0$$

Crack Initiation Model



Exercise

- To optimize a given pavement AC thickness against top down cracking

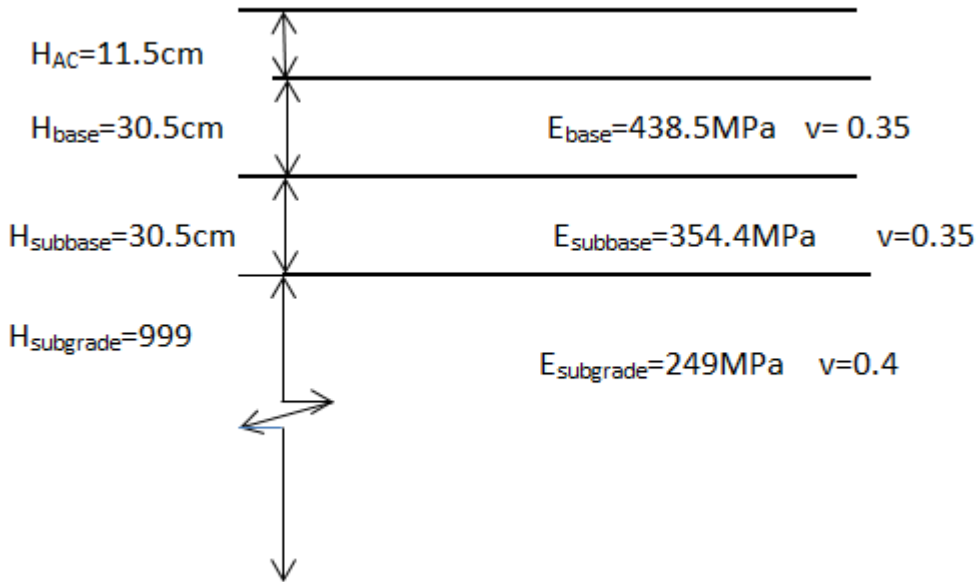


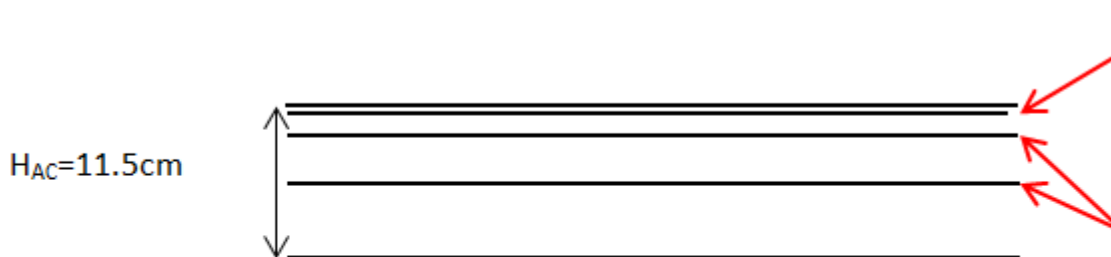
Figure 1 Pavement system

1) Input all required parameters

2) Determine ER_{opt} using ESALs & reliability

4) Calculate the $DCSE_{lim}$

6) Determine the optimized AC thickness



3) Determine creep parameters and tensile strength at $z = 0.25 \text{ inch}$

5) Determine AC stiffness at $h = 0.125 * h_{ac}$ and $h = 0.5 * h_{ac}$

Figure 2 AC layer

END

Questions

